

# The Effect of Sequence of Operations on Fatigue Life of LSP Treated Open-hole Aluminium Specimens



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# Outline

- Introduction
- LSP treatment
- Residual stress measurements
- Fatigue testing
- Conclusions

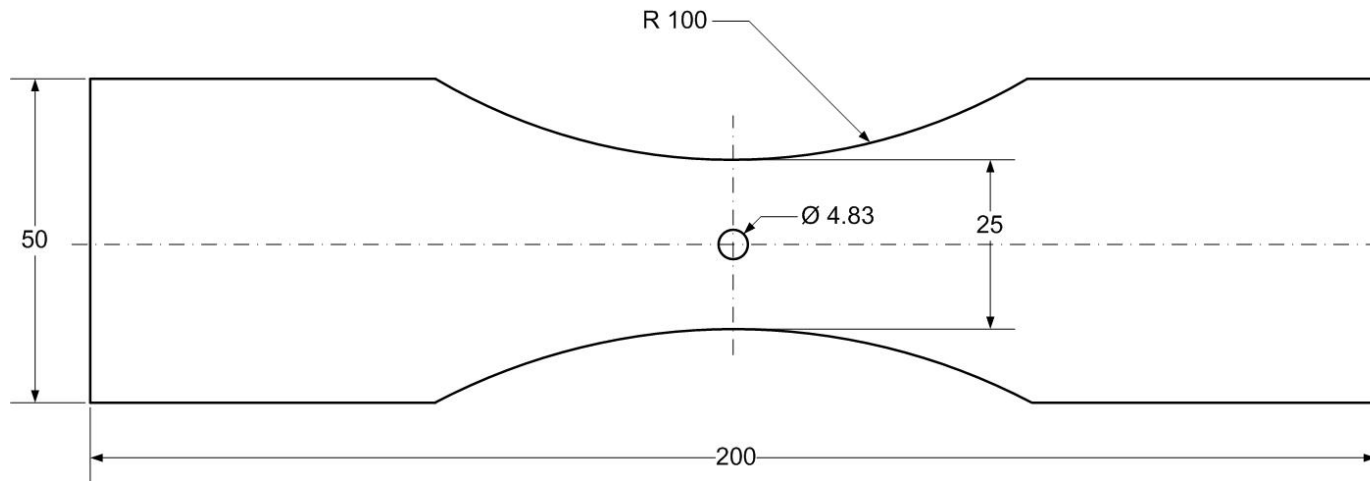
# Introduction

- Previous work\* on LSP treatment on thin, open-hole specimens resulted in necessity to optimize the process
- Observed reduced performance of LSP - the treatment was performed on the specimens with an open-hole already present
- An additional experimental campaign - to highlight the importance of the sequence of operations: hole drilling and LSP treatment

\*G. Ivetic, I. Meneghin, E. Troiani, G. Molinari, A. Lanciotti, V. Ristori, J.L. Ocaña, M. Morales, J.A. Porro, C. Polese, A.M. Venter, "Characterisation of Fatigue and Crack Propagation in Laser Shock Peened Open Hole 7075-T73 Aluminium Specimens" , ICAF 2011, Structural Integrity: Influence of Efficiency and Green Imperatives: Proceedings of the 26th Symposium of the International Committee on Aeronautical Fatigue

# Specimens@Bologna

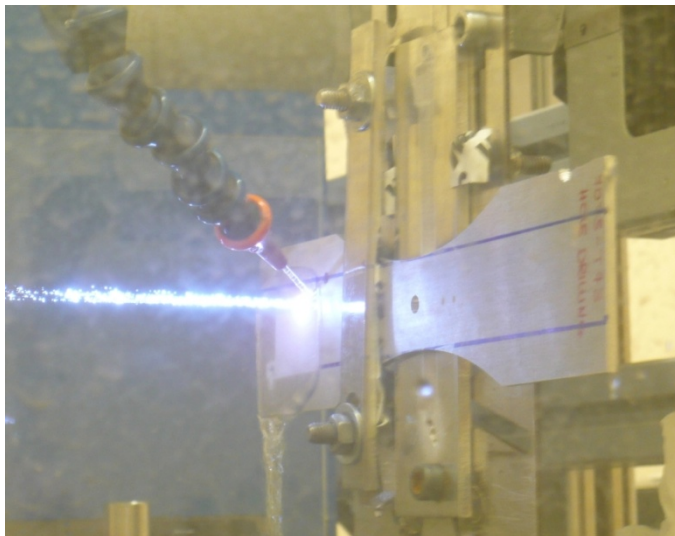
- Dog-bone specimens in AA 6082-T6
- CNC cut from a 3 mm thick sheet



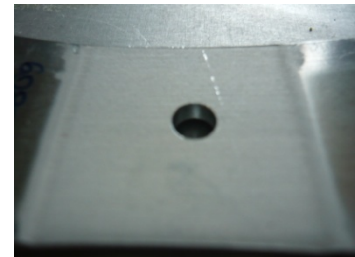
# LSP Treatment@Madrid

- Process settings

Laser type	Wavelength [nm]	Output energy [J]	Pulse duration [ns]	Peen size [mm]	Laser frequency [Hz]
Nd-YAG	1064	2.8	9	1.5	10



- Peen density of 625 peens per cm square
- Two side peening, one side at the time
- Treated area – zone around the open-hole, size 25x25 mm
- Specimen fixing – two point encastre

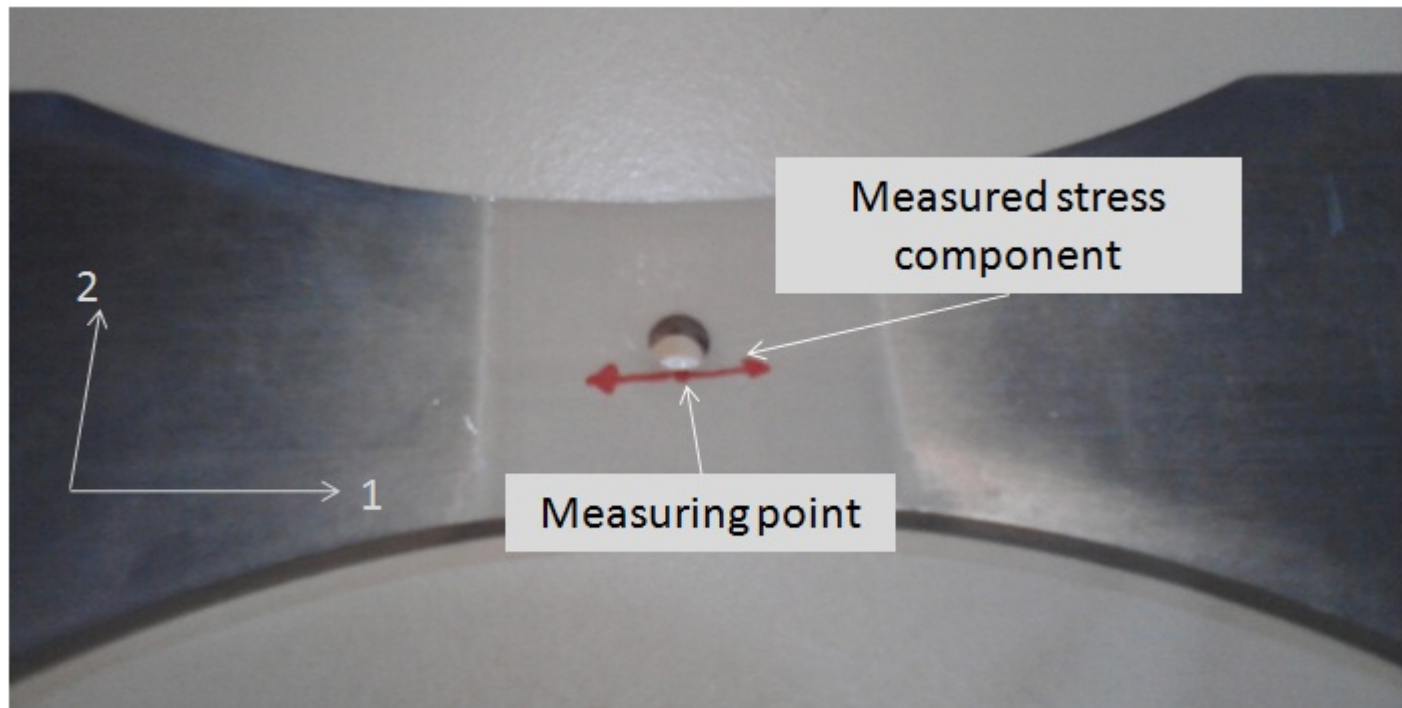


# Residual stress measurements@Elettra

- Elettra Synchrotron facility, MCX Beamline, Trieste, Italy
- The residual stresses were measured at the edge of the open hole, for the two different types of specimens (specimen LSP treated with pre-existing open hole, i.e. hole before, and specimen with the hole drilled on the LSP treated area, i.e. hole after)

# Residual stress measurements@Elettra

Depth [ $\mu\text{m}$ ]	Residual stress – hole before [MPa]	Residual stress – hole after [Mpa]
50	-76	-127
125	-46	-129
250	-117	-175



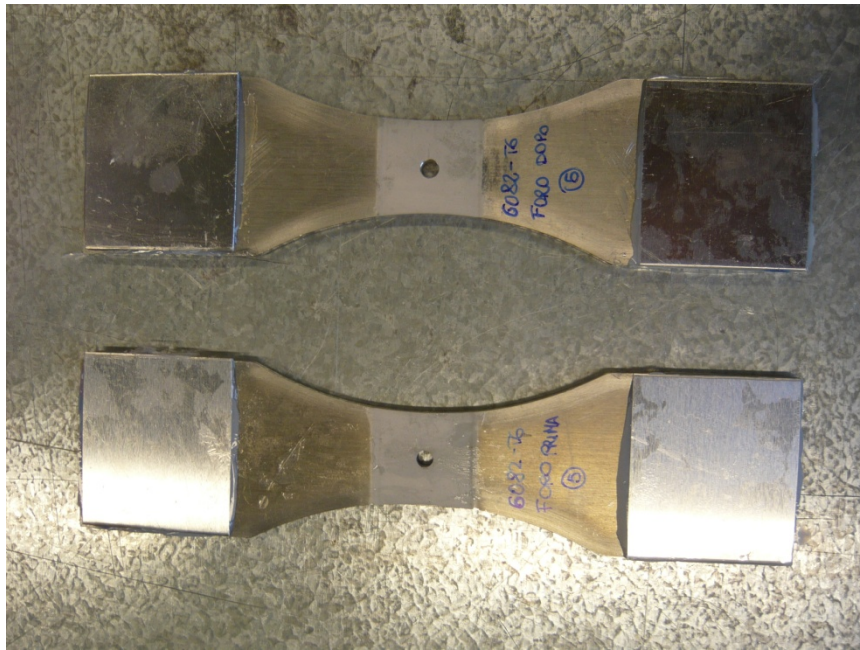
# Residual stress measurements@Elettra

- Amount of compressive residual stresses is bigger in the case where the open hole is drilled after the LSP treatment
- This gives first indications on the positive effects introduced by drilling the hole only after the LSP treatment has been made



# Fatigue testing@Bologna

- Additional tabs were bonded to the clamped sides of the specimens in order to avoid fretting fatigue damage



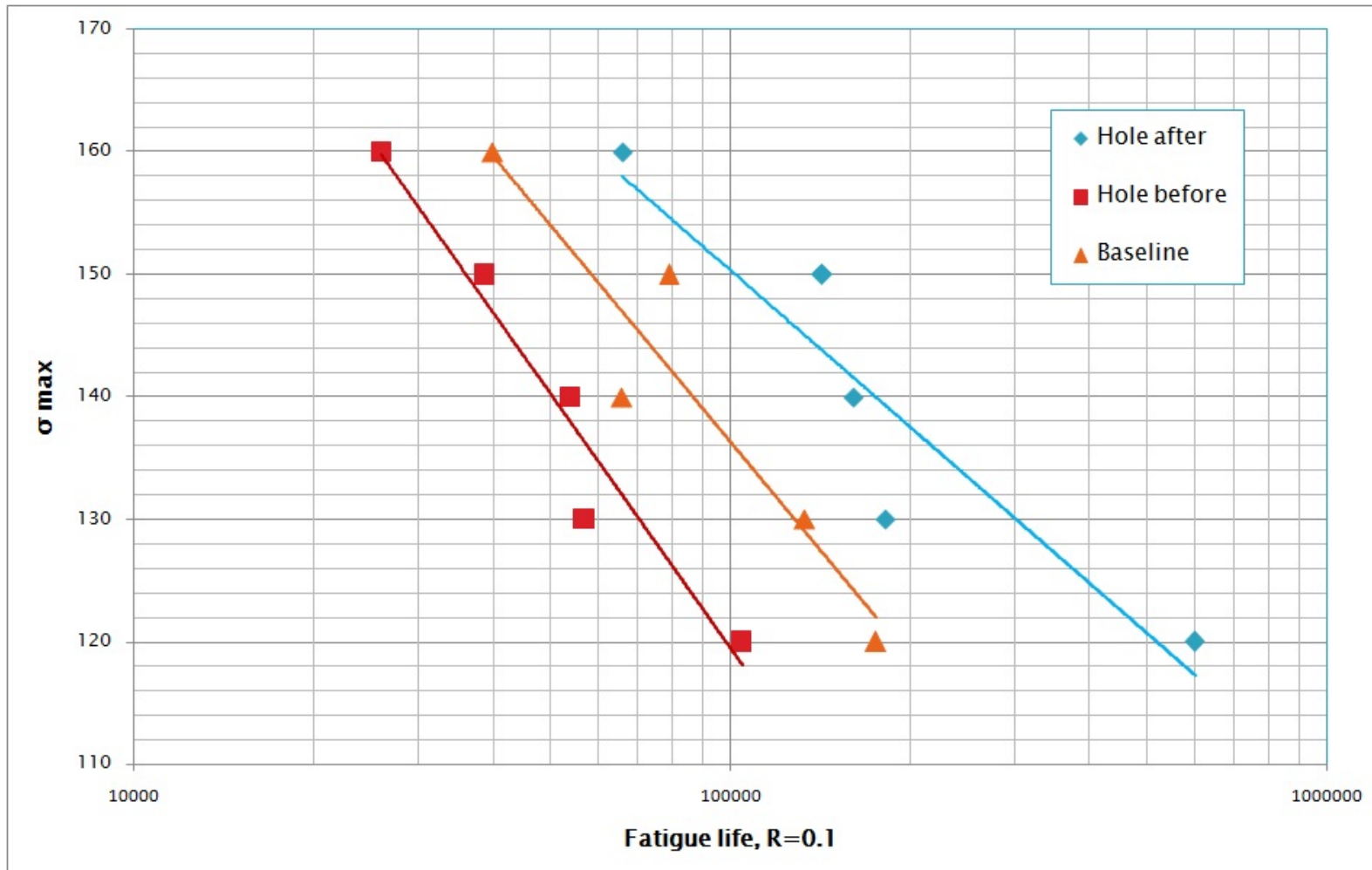
# Fatigue testing@Bologna

- Maximum nominal stresses ranging from 120 MPa to 160 MPa
- $R=0.1$



# Fatigue testing@Bologna

A total of 15 specimens was tested; five for each case (hole before, hole after, baseline)



# Fatigue testing@Bologna

Fatigue lives with different operation sequences at R=0.1 (BL=Baseline)

$\sigma_{max}$ MPa	BL ( $10^4$ cycles)	Hole before	Hole before/BL	Hole after	Hole after/BL
120	17.54	10.46	0.59	60.04	3.42
130	13.32	5.68	0.42	18.16	1.36
140	6.58	5.41	0.82	16.08	2.44
150	7.87	3.87	0.49	14.25	1.80
160	3.99	2.62	0.65	6.63	1.65

# Results

- Drilling the hole after the LSP treatment can indeed increase fatigue lives of treated specimen
- The practical implications - a good solution for "in production" application, in which holes are to be drilled after the LSP treatment
- The "in service" application has proven to be impracticable due to negative effects on fatigue lives that were encountered on the specimens with the hole already present

# Conclusions

- LSP needs to be optimized for every application used, specially when it comes to low thickness specimens
- The experiments in the coating-free LSP configuration have shown that the sequence of operations plays a crucial role in increasing the fatigue life of the treated specimens

# Conclusions

- If the hole is realized after the LSP treatment is performed on the specimen, the fatigue life increase is up to three times more than the life of the baseline specimens
- If the hole is realized before the LSP treatment is performed on the specimen, the fatigue life is about two times less than the life of the baseline specimens
- Results submitted to Material Science and Engineering A

# Future work

- Hypothesis – the presence of tensile residual stresses at the mid-section of the “hole before” specimen as the cause of premature fatigue failure (not measured experimentally due to low penetration of used synchrotron radiation)
- Scanning electron microscope investigation of fracture surfaces in order to confirm hypothesis of crack nucleation at the mid-thickness of the “hole before “ specimens



Thank you for your attention!



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